

# ON SOME PRACTICAL ISSUES IN SYSTEM IDENTIFICATION FOR THE WINDSURFER APPROACH TO ADAPTIVE ROBUST CONTROL

W.S. LEE\*, B.D.O. ANDERSON\*, I.M.Y. MAREELS\*, R.L. KOSUT \*\*

\*Department of Systems Engineering and Cooperative Research Centre for Robust and Adaptive Systems, Australian National University, Canberra, ACT 0200, AUSTRALIA

\*\*Integrated Systems Inc., 9260 Jay St., Santa Clara, CA 95054, USA

**Abstract.** In this paper we investigate some important issues in system identification for the windsurfer approach to robust adaptive control. In particular, correlation function estimates and power spectrum estimates are compared as methods for model validation. This comparison leads us to suggest a reliable procedure for deciding (1) when should we identify a better model, and (2) whether we have identified a good model for our purposes.

**Key Words.** Adaptive control; control system design; identification; model reference control; robustness

## 1. INTRODUCTION

### 1.1. Background and Objectives of This Paper

Iterative identification and control design is a topic of growing interest recently (see, for example, Anderson and Kosut (1991), Lee *et al.* (1993a), Partanen and Bitmead (1993), Schrama and den Hof (1992), and Zang *et al.* (1991)). One of the schemes proposed in this area is the windsurfer approach which attempts to increase the bandwidth of a closed-loop system while keeping the closed-loop frequency response approximately flat in the passband, given that the initial model of the plant may involve significant error in the high-frequency region. It was demonstrated recently in Lee *et al.* (1993b) that when the performance robustness of the closed-loop system deteriorates (while the closed-loop bandwidth is increased to a certain value), it is possible to improve the performance robustness of the closed-loop system through an iterative identification and controller design procedure.

In this paper we shall investigate some important issues in system identification for the windsurfer approach. In particular we would like to know (1) when should we identify a better model, and (2) whether we have identified a good model for our purposes.

### 1.2. Structure of The Paper

To establish notation and relevant equations, we describe in Section 2 the system identification method employed by the windsurfer approach. Necessary conditions for identifying a good model are also outlined. In Section 3 we present a frequency domain method for model validation. In Section 4 we describe a time domain method for model validation. In Section 5 the model validation methods are compared through case studies by simulations. In Section 6 we explain how the model validation methods can provide a reliable procedure for answering the questions highlighted at

the end of Subsection 1.1.

## 2. CONTROL-RELEVANT SYSTEM IDENTIFICATION

Given a stable strictly proper model  $G_0$  of a stable strictly proper plant  $G$ , it was shown in Lee *et al.* (1993b) that we can design a sequence of controllers  $\{K_0^j\}$  such that the designed closed-loop transfer functions

$$\bar{T}_0^j = \frac{G_0 K_0^j}{1 + G_0 K_0^j}$$

are well behaved and the sequence  $\{\bar{T}_0^j\}$  has progressively increasing bandwidth  $\lambda_0^j$ . The performance robustness of the actual closed-loop transfer function,

$$T_0^j = \frac{G K_0^j}{1 + G K_0^j},$$

is measured by

$$J = \|v\|_2^2 \quad (1)$$

where

$$v = (T_0^j - \bar{T}_0^j) r; \quad r = \text{reference input}, \quad (2)$$

is the tracking error induced by the model error associated with  $G_0$  under closed-loop conditions.

**Remark**

- Note that the tracking error  $v$  cannot be measured directly. It can only be estimated from the closed-loop output error  $\xi$  shown in Fig. 1. It can be easily seen that

$$\xi = v + w \quad (3)$$

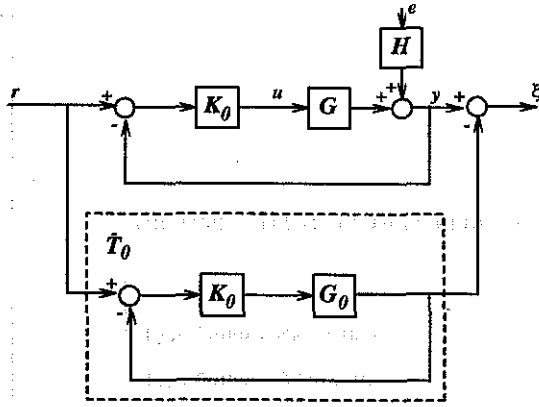


Fig. 1. Closed-loop output error  $\xi$

where  $w = (1 + GK_0^f)^{-1}He$  is the effect of the noise,  $e$ , on the actual closed-loop output.

When  $\lambda_0^j$  is increased to a certain value,  $\lambda_0^f$ ,  $J^f = \|v^f\|_2^2$  (the value of  $J$  when  $\lambda_0^j = \lambda_0^f$ ) may become excessively large. At this stage, we like to identify a new model  $G_1$  such that

$$\left\| \left( \frac{GK_0^f}{1 + GK_0^f} - \frac{G_1K_0^f}{1 + G_1K_0^f} \right) r \right\|_2^2$$

is minimized. Note that  $K_0^f$  is designed on the basis of  $G_0$ , when the value of  $\lambda_0^j$  is  $\lambda_0^f$ .

This closed-loop identification problem can be transformed into an open-loop identification problem by employing Hansen's framework of identification (Hansen, 1989). We quote a special case (where the plant and its models are stable) of Theorem 2 proved in Lee *et al.* (1993a).

*Theorem 1* Let  $K_0^f = (1 - G_0Q_0^f)^{-1}Q_0^f$  stabilize  $G$  and  $G_0$ , where  $Q_0^f$  is a proper stable transfer function.  $G$  can be parametrized by a strictly proper stable transfer function  $R_0^f$  via

$$G = G_0 + \frac{R_0^f}{1 - R_0^fQ_0^f}$$

Let

$$G_1 = G_0 + \frac{\hat{R}_0^f}{1 - \hat{R}_0^fQ_0^f} \quad (4)$$

be another model stabilized by  $K_0^f$ , where  $\hat{R}_0^f$  is a strictly proper stable estimate of  $R_0^f$ .

Also define

$$\xi_1 = (1 - \bar{T}_0^f)(\beta - \hat{R}_0^f\alpha) \quad (5)$$

where  $\alpha = Q_0^f r$ ,  $\beta = y^f - G_0 w^f$ , and  $w^f$  and  $y^f$  are, respectively, the input and output of the plant resulting from the application of  $K_0^f$ . Then  $\xi_1$  can be expressed as

$$\xi_1 = \left( \frac{GK_0^f}{1 + GK_0^f} - \frac{G_1K_0^f}{1 + G_1K_0^f} \right) r + w^f \quad (6)$$

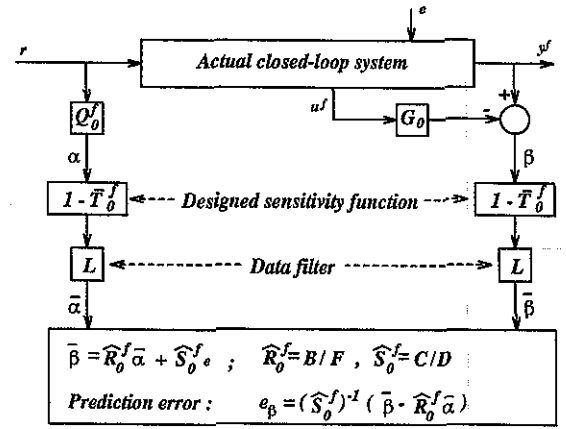


Fig. 2. Identification of  $R_0^f$  and  $S_0^f$

where  $w^f = (1 + GK_0^f)^{-1}He$ .

Remarks

- If we define  $H = (1 - R_0^fQ_0^f)^{-1}S_0^f$ , where  $S_0^f$  is a proper stable and inversely stable transfer function, then the actual closed-loop system has Hansen's open-loop representation

$$\beta = R_0^f\alpha + S_0^fe \quad (7)$$

- Since the "input"  $\alpha$  in equation (7) is independent of the noise disturbance  $e$ , identifying  $R_0^f$  (and  $S_0^f$ ) is an *open-loop* identification problem.
- We shall identify  $R_0^f$  and  $S_0^f$  by the prediction error method (Ljung, 1987) as shown in Fig. 2. Data filters  $L$  (typically low-pass) are usually employed to shape the bias-distribution of the estimates (which is due to undermodelling) such that the model error is small in the appropriate frequency range.

### 3. A FREQUENCY DOMAIN METHOD FOR MODEL VALIDATION

In any system identification problem, it is important to ensure that a model with the right properties is identified. In the following we shall present a model validation method in the frequency domain. It should be emphasized that the model validation procedure is designed with the closed-loop control objective in mind.

Recall that, given the existing model  $G_0$ , it is necessary to identify an improved model  $G_1$  when  $J^f = \|v^f\|_2^2$  is excessively large. Evidently  $\xi^f$  could be large (implying undesirable performance) with one of both of  $v^f$  and  $w^f$  large. If the former is larger, there is a potential to reduce it by improved model identification. But this will only work (in a particular frequency band) if the signal-to-noise ratio is sufficiently high. Specifically, when only finite durations of input-output measurements are available for identifying  $R_0^f$  (which parametrizes  $G_1$ ), it was shown in Lee *et al.* (1993b) that the normalized variance of  $\hat{R}_0^f$  will be small only if the signal-to-noise ratio,  $\Phi_{v^f}(\omega)/\Phi_{w^f}(\omega)$ , associated with  $\xi^f = v^f + w^f$  is sufficiently high, where  $\Phi_{v^f}(\omega)$  and  $\Phi_{w^f}(\omega)$  are, respectively, the power spec-

tra of  $v^f$  and  $w^f$ . Obviously, then one needs to estimate power spectra for  $w$  and  $v$  (or more precisely  $\xi$ ). We shall proceed as follows.

From equations (2) and (3) we observe that when  $r = 0$ , the sole contributor to  $\xi$  is  $w$ . Therefore we can compute  $\Phi_w(\omega)$  after measuring  $\xi$  with  $r = 0$ . When  $r \neq 0$ , we have  $\xi = v + w$ . Assuming that  $v$  and  $w$  are uncorrelated (which follows if  $r$  and  $e$  are uncorrelated, a typical situation), then  $\Phi_\xi(\omega) = \Phi_v(\omega) + \Phi_w(\omega)$ . By visual comparison of  $\Phi_\xi(\omega)$  with  $\Phi_w(\omega)$ , we evaluate the significance of  $\Phi_v(\omega)$  with respect to  $\Phi_w(\omega)$ . If  $\Phi_{\xi^f}(\omega)$  is significantly larger than  $\Phi_{w^f}(\omega)$  in a frequency band spanning one decade and centered around  $\lambda_0^f$  (when the designed closed-loop bandwidth is  $\lambda_0^f$ ), the model  $G_0$  is invalidated for the design of closed-loop systems with bandwidths larger than or equal to  $\lambda_0^f$ .

The method just described can also be used to validate  $G_1$  after it is identified. We simply replace  $G_0$  by  $G_1$ , while retaining  $K_0^f$ , in the simulation of the designed closed-loop response to the reference input. This allows us to compute  $\xi_1$  and its power spectrum  $\Phi_{\xi_1}(\omega)$ . By visually comparing  $\Phi_{\xi_1}(\omega)$  with  $\Phi_w(\omega)$ , we have good confidence that  $G_1$  is a reliable model of  $G$  (when the designed closed-loop bandwidth is  $\lambda_0^f$ ) if  $\Phi_{\xi_1}(\omega)$  is comparable to  $\Phi_w(\omega)$  up to  $\lambda_0^f$ .

#### 4. A TIME DOMAIN METHOD FOR MODEL VALIDATION

We shall now describe a time domain model validation method. This is useful both for establishing that  $G_0$  should be rejected (that is, as a flag for re-identification) as well as for validating a new model,  $G_1$ , replacing  $G_0$ .

Referring to Fig. 2 and equation (5), we notice that  $e_\beta = L\xi_1$  when  $\hat{S}_0^f = 1$ , where  $e_\beta$  is the prediction error (also known as the residual). We also observe from equation (4) that  $G_1 = G_0$  when  $\hat{R}_0^f = 0$ . Equation (6) shows that  $\xi_1 = \xi^f$  when  $G_1 = G_0$ . Therefore we have  $e_\beta = L\xi^f$  when  $\hat{S}_0^f = 1$  and  $\hat{R}_0^f = 0$ . This suggests that  $G_0$  should be rejected if the cross-correlation of the prediction error  $e_\beta$  with the future values of "input"  $\bar{\alpha}$  exceed its ( $3\sigma$ ) confidence limits when  $\hat{S}_0^f = 1$  and  $\hat{R}_0^f = 0$ . This reasoning is independent of the true  $S_0^f$ . See Ljung (1987) for more details of model validation by correlation techniques. (Actually it is also easy to apply the same method to validate a pair of newly identified  $\hat{R}_0^f$  and  $\hat{S}_0^f$  before  $\hat{R}_0^f$  is used to calculate  $G_1$ . We simply check that the auto-correlation function of  $e_\beta$  for non-zero delays as well as the cross-correlation of  $e_\beta$  with the future values of  $\bar{\alpha}$  are within their respective confidence intervals.)

#### 5. COMPARING THE MODEL VALIDATION METHODS

Sections 3 and 4 presented two different methods for model validation. In the following, we shall compare the two methods in two simulation examples. In these examples, the plant is a simulated flexible link robot arm whose transfer function  $G$  has poles at  $-0.0996 \pm$

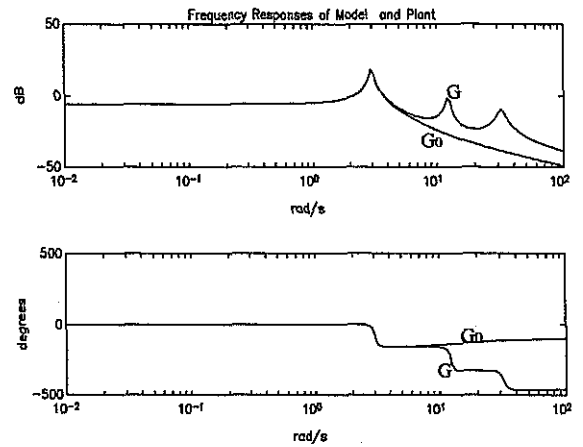


Fig. 3. Frequency response of model  $G_0$

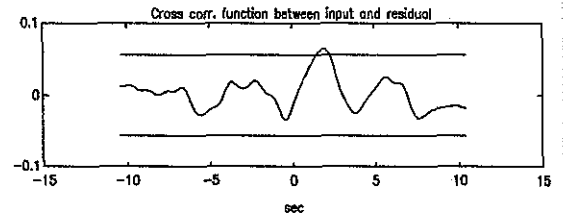


Fig. 4. Validating  $G_0$  ( $\lambda_0^f = 1.5 \text{ rad/s}$ )

$j3.0017$ ,  $-0.3339 \pm j12.131$ ,  $-1.845 \pm j31.481$ , zeros at  $s = -13.162$ ,  $-10.646 \pm j12.27$ ,  $s = 7.169 \pm j11.54$ , and  $G(0) = 0.5196$ .

#### 5.1. Example One

The initial model  $G_0$  is an open-loop description of  $G$  up to and including its first resonant frequency (see Fig. 3).  $G_0$  has a pair of poles at  $-0.0903 \pm j3.0027$ , a zero at  $s = -13.31$ , and  $G_0(0) = 0.5188$ .

When the designed closed-loop bandwidth is  $1.5 \text{ rad/s}$ , the method of correlations (see Fig. 4) shows that  $G_0$  is not a good model of  $G$  whereas the method of power spectra (compare Fig. 5 and 6) shows that the tracking error is still insignificant. In fact, we are unable to identify a better model than  $G_0$  at this stage. When the designed closed-loop bandwidth has increased to  $3 \text{ rad/s}$ , the method of correlations (see Fig. 7) and the method of power spectra (compare Fig. 8 and 9)

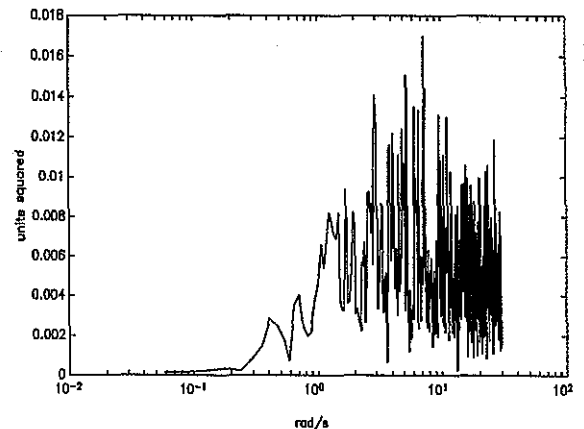


Fig. 5.  $\Phi_w(\omega)$  when  $\lambda_0^f$  is  $1.5 \text{ rad/s}$

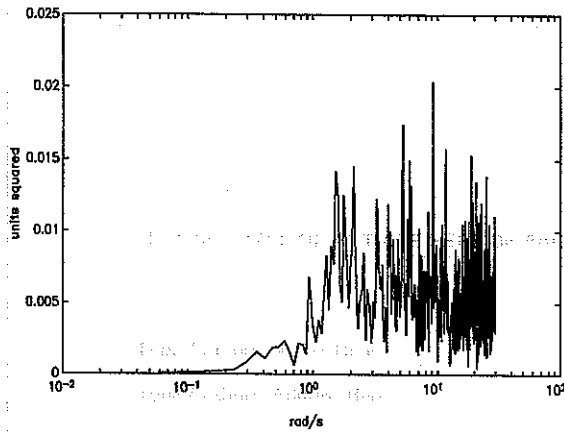


Fig. 6.  $\Phi_{\xi}(\omega)$  when  $\lambda_0^f$  is 1.5 rad/s

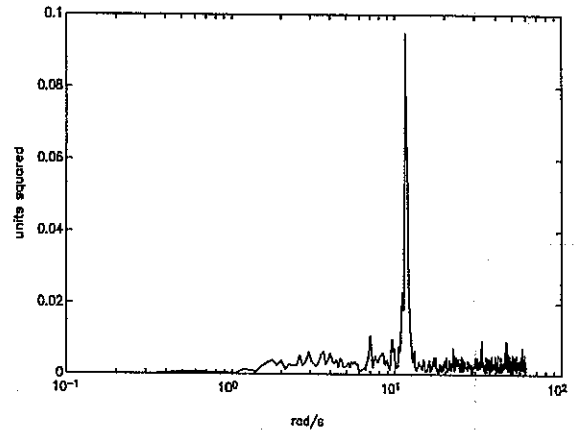


Fig. 9.  $\Phi_{\xi}(\omega)$  when  $\lambda_0^f$  is 3 rad/s

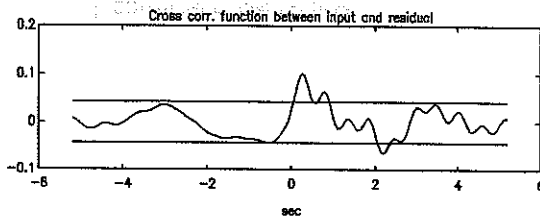


Fig. 7. Validating  $G_0$  ( $\lambda_0^f = 3$  rad/s)

both indicate that  $G_0$  is not a good model. In particular, comparison of Fig. 8 and 9 indicates that the closed-loop output error has a high signal-to-noise ratio at around 12 rad/s. The identified  $\hat{R}_0^f$  and  $\hat{S}_0^f$  are validated by the method of correlations (see Fig. 10) before  $\hat{R}_0^f$  is used to calculate  $G_1$ . The resulting  $G_1$  is validated by the method of power spectra (compare Fig. 11 and 8). Fig. 12 shows the frequency response of  $G_1$ .

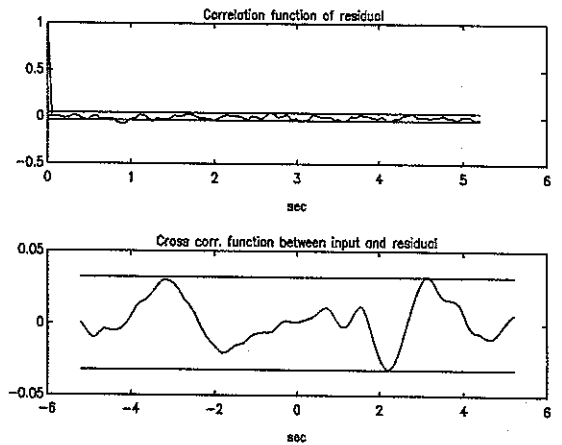


Fig. 10. Validating  $\hat{R}_0^f$  and  $\hat{S}_0^f$  ( $\lambda_0^f = 3$  rad/s)

## 5.2. Example Two

By employing the model  $G_1$  of example 1 (which has poles at  $s = -0.0895 \pm j3.0026, -0.4834 \pm j12.03, -2.475 \pm j31.502$ , zeros at  $-12.967, -7.336 \pm j11.05, 9.098 \pm j12.07$ , and  $G_1(0) = 0.5189$ ), it is possible to increase the designed closed-loop bandwidth to 12 rad/s before it is necessary to identify a better model. At this stage, the method of correla-

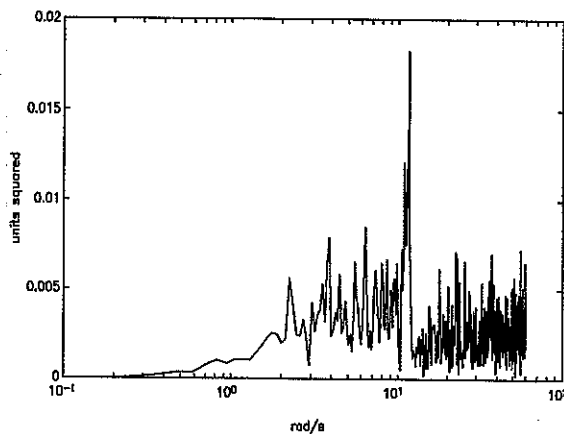


Fig. 8.  $\Phi_w(\omega)$  when  $\lambda_0^f$  is 3 rad/s

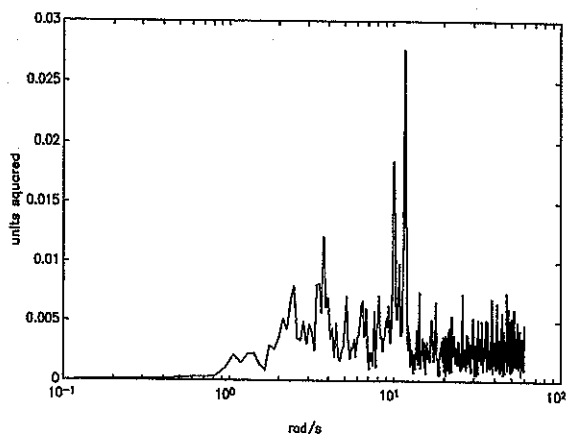


Fig. 11.  $\Phi_{\xi_1}(\omega)$  when  $\lambda_0^f$  is 3 rad/s

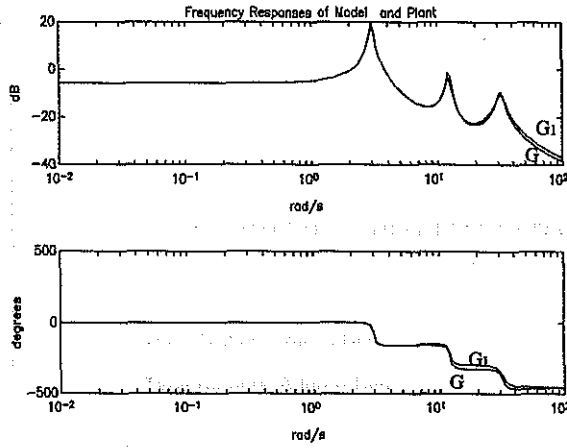


Fig. 12. Frequency response of model  $G_1$

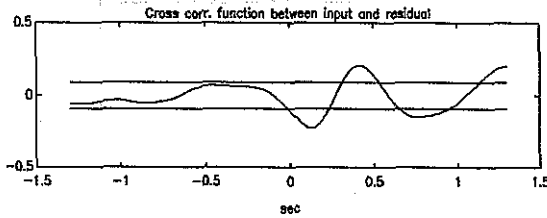


Fig. 13. Validating  $G_1$  ( $\lambda_1^f = 12 \text{ rad/s}$ )

tions (see Fig. 13) and the method of power spectra (compare Fig. 14 and 15) both indicate that  $G_1$  is not a good model. However, we are unable to identify a better model despite considerable efforts and numerous attempts. This phenomenon can be explained as follows. Notice that the designed closed-loop bandwidth ( $12 \text{ rad/s}$ ) is close to the critical frequency corresponding to the *unstable zeros* of  $G_1$  (at  $s = 9.098 \pm j12.07$ ). From the view point of control design, it follows from Freudenberg and Looze (1985) that, due to the unstable zeros of  $G_1$ , the system has reached its fundamental performance limitations. Furthermore, it can be shown that the tracking error and the effect of noise disturbance at this stage are, respectively, given by

$$v^f = \frac{\left(\frac{G-G_1}{G_1}\right) \bar{T}_1^f}{1 + \left(\frac{G-G_1}{G_1}\right) \bar{T}_1^f} (1 - \bar{T}_1^f) r ,$$

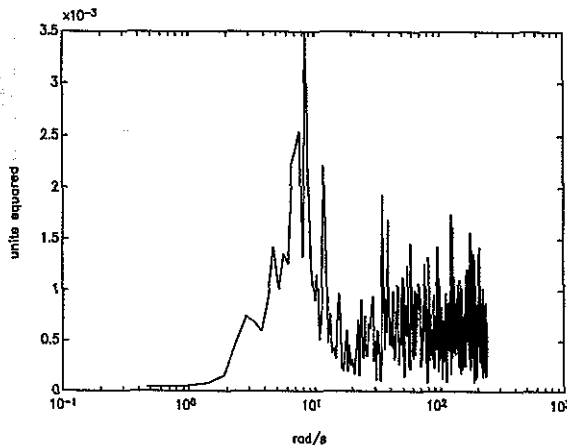


Fig. 14.  $\Phi_w(\omega)$  when  $\lambda_1^f$  is  $12 \text{ rad/s}$

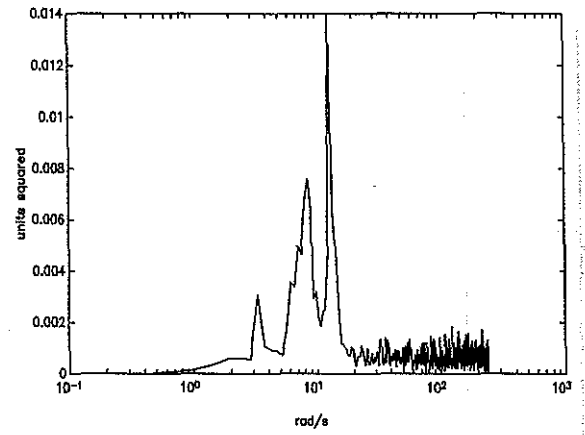


Fig. 15.  $\Phi_\xi(\omega)$  when  $\lambda_1^f$  is  $12 \text{ rad/s}$

and

$$w^f = \frac{1}{1 + \left(\frac{G-G_1}{G_1}\right) \bar{T}_1^f} (1 - \bar{T}_1^f) H e ,$$

where

$$\bar{T}_1^f = \left(\frac{12}{s + 12}\right) [G_1]_a$$

is the designed closed-loop transfer function with

$$[G_1]_a = \text{the all-pass factor associated with } G_1 .$$

Note that for the case where  $G_1$  has no unstable zeros,  $|1 - \bar{T}_1^f(j\omega)| \leq 1$  for all frequencies. However, in the presence of the unstable zeros of  $G_1$  around  $12 \text{ rad/s}$  and when the designed closed-loop bandwidth is close to  $12 \text{ rad/s}$ ,  $|1 - \bar{T}_1^f(j\omega)| \approx 2$  at around  $12 \text{ rad/s}$ . Therefore the tracking error  $v^f$  and the effect of noise disturbance  $w^f$  are both amplified by a factor of approximately two at around  $12 \text{ rad/s}$  in the latter situation. Hence, for the same magnitude of  $[(G - G_1)/G_1] \bar{T}_1^f$  in the neighbourhood of  $12 \text{ rad/s}$ , the value of  $\Phi_{\xi^f}(\omega)$  for the case where  $G_1$  has unstable zeros at around  $12 \text{ rad/s}$  is about four times of that for the case where  $G_1$  has no unstable zeros. However, since  $\Phi_{v^f}(\omega)/\Phi_{w^f}(\omega)$  does not depend on  $1 - \bar{T}_1^f$ , the increase in  $\Phi_{\xi^f}(\omega)$  that we have just described does not increase the signal-to-noise ratio of  $\xi^f$ . This implies that we are caught in a situation where it is necessary to improve the model (because  $\Phi_{\xi^f}(\omega)$  is excessively large near to  $12 \text{ rad/s}$ ) but we are unable to identify a better model (because  $\Phi_{v^f}(\omega)/\Phi_{w^f}(\omega)$  is not sufficiently high). We therefore conclude that, owing to the controller designed on the basis of  $G_1$  for a closed-loop bandwidth of  $12 \text{ rad/s}$ , the closed-loop identification process has reached its natural limits of accuracy constrained by the unstable zeros of  $G_1$  in the region of  $12 \text{ rad/s}$ .

### 5.3. Lessons Learned from The Examples

From the examples and many other simulation studies, we can make the following conclusions:

- Correlation function estimates and power spectrum estimates are both useful for model vali-

dation where the goodness of fit is based on a closed-loop control criterion.

- Correlation function estimates are more sensitive than power spectrum estimates in the sense that the former tend to invalidate a model before identifying a better model is necessary and possible. This *does not imply* that the correlation method is useless. On the contrary, it suggests that the correlation method is useful for detecting incipient model errors.
- Power spectrum estimates not only suggest when a model becomes inadequate but they also indicate the frequency range in which the signal-to-noise ratio is high for identification.
- Prior knowledge of unstable zeros in the existing model (say  $G_0$ ) is important. As explained in Example two, there is a fundamental limit of accuracy for performing identification on closed-loop systems involving controllers designed on the basis of models with unstable zeros. Let  $\omega_z$  be the minimum critical frequency corresponding to the unstable zeros of  $G_0$ , simulation evidence suggests that it is very difficult, if not impossible, to identify a model better than  $G_0$  if  $\lambda_0^f \geq \omega_z/2$ . It should be remarked that this is reminiscent of design tradeoffs discussed in Freudenberg and Looze (1985), as opposed to an ill-posed problem.
- In general we should update  $G_0$  if and only if
  1. both methods of model validation suggest that it is necessary to do so, *and*
  2.  $\lambda_0^f < \omega_z/2$ , where  $\omega_z$  is the minimum critical frequency corresponding to the unstable zeros of  $G_0$ .

## 6. IDENTIFICATION OF A BETTER MODEL

We have remarked in section 2 that under practical conditions, the accuracy of the identified model can be improved by increasing the signal-to-noise ratio of the closed-loop output error. This is equivalent to having a certain level of deterioration in robust performance relative to the effect of noise disturbance. However, it is undesirable from the control point of view for robust performance to deteriorate too seriously. In this section, we explain how these competing objectives can be resolved.

The signal-to-noise ratio of the closed-loop output error in the frequency range where the current model  $G_0$  has significant error can be increased by increasing the magnitude of the reference input or by increasing the designed closed-loop bandwidth. If practical operation constraints do not allow the magnitude of the reference input to be increased, then the signal-to-noise ratio of the closed-loop output error can only be increased by increasing the designed closed-loop bandwidth. This, however, has the potential danger of causing instability in the actual closed-loop system if the designed closed-loop bandwidth is increased excessively. To avoid this danger, we would like to suggest the following procedure:

1. Reduce the rate of increasing the designed closed-loop bandwidth  $\lambda_0^j$  once the correlation method for model validation has invalidated  $G_0$ .
2. Attempt to identify  $R_0^f$  (when  $\lambda_0^j = \lambda_0^f$ ) as soon

as the power spectrum method for model validation suggests that  $\xi^f$  has a sufficiently high signal-to-noise ratio, provided that  $\lambda_0^f < \omega_z/2$ , where  $\omega_z$  is the minimum critical frequency corresponding to the unstable zeros of  $G_0$ .

- (a) Use the collected data to identify a set of models by experimenting with the likely model structures. Perform model verification on each of these models.
  - (b) If an identified model is found to be sufficiently accurate, accept it for the next stage of control design. Otherwise, increase the designed closed-loop bandwidth slightly, collect a new set of measurements and repeat the procedures of model estimation and verification.
  - (c) Repeat the last two steps until a sufficiently accurate model is obtained and verified.
3. Terminate the iterative identification and control design procedure if  $\lambda_0^f \geq \omega_z/2$  and  $\xi^f$ , although unacceptably large, does not facilitate the identification of a better model.

## 7. ACKNOWLEDGEMENTS

The authors wish to acknowledge the funding of the activities of the Cooperative Research Centre for Robust and Adaptive Systems by the Australian Government under the Cooperative Research Centres Program. W. S. Lee wishes to thank Dr Robert R. Bitmead for his valuable advice.

## 8. REFERENCES

- Anderson, B.D.O. and R.L. Kosut (1991). Adaptive robust control: On-line learning. *Proc. CDC'91* pp. 297-298.
- Freudenberg, J.S. and D.P. Looze (1985). Right half plane poles and zeros and design tradeoffs in feedback systems. *IEEE Trans. Automat. Contr.* **30**, 555-565.
- Hansen, F.R. (1989). *Approach to Closed-loop System Identification and Experiment Design (Ph.D. Dissertation)*. Stanford University. Stanford, CA, USA.
- Lee, W.S., B.D.O. Anderson, R.L. Kosut and I.M.Y. Mareels (1993a). A new approach to adaptive robust control. *Int. J. Adaptive Control and Signal Processing* **7**, 183-211.
- Lee, W.S., B.D.O. Anderson, R.L. Kosut and I.M.Y. Mareels (1993b). On robust performance improvement through the windsurfer approach to adaptive robust control. *Proc. CDC'93* pp. 2821-2827.
- Ljung, L. (1987). *System Identification: Theory for the user*. Prentice-Hall. Englewood Cliffs.
- Partanen, A.G. and R.R. Bitmead (1993). Two stage iterative identification/control design and direct experimental controller refinement. *Proc. CDC'93* pp. 2833-2838.
- Schrama, R.J.P. and P.M.J. Van den Hof (1992). An iterative scheme for identification and control design based on coprime factorizations. *Proc. ACC'92* pp. 2842-2846.
- Zang, Z., R.R. Bitmead and M. Gevers (1991). Iterative model refinement and control robustness enhancement. *Proc. CDC'91* pp. 279-284.